

# **An Impact Sensor System for the Characterization of the Micrometeoroid and Lunar Secondary Ejecta Environment**

**J.-C. Liou<sup>(1)</sup>, M. Burchell<sup>(2)</sup>, R. Corsaro<sup>(3)</sup>, F. Giovane<sup>(4)</sup>,  
E. Stansbery<sup>(1)</sup>, J. Blum<sup>(5)</sup>, W. Cooke<sup>(6)</sup>, and V. Pisacane<sup>(7)</sup>**

**<sup>(1)</sup>NASA/JSC, <sup>(2)</sup>Kent, <sup>(3)</sup>NRL, <sup>(4)</sup>VT, <sup>(5)</sup>TU-BS, <sup>(6)</sup>NASA/MSFC, <sup>(7)</sup>USNA**

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## Outline

- **Project objectives**
- **Lunar particle environment**
- **Benefits**
- **System overview**
- **Current status**
- **Plan forward**



## Objectives

- **Short-term: Conduct Pre-Phase A and Phase A activities of two integrated micrometeoroid and lunar secondary ejecta (MMSE) impact detectors**
  - Impact Sensors for Micromeeteoroid and Lunar Secondary Ejecta (IMMUSE)
  - Funded by the NASA LASER Program through 2011
- **Long-term: Conduct MMSE *in situ* measurements on the Moon to better define the particle environment for future lunar exploration activities**



## Lunar Surface Environment (1/3)

- **The Moon continues to get bombarded by meteoroids from space**



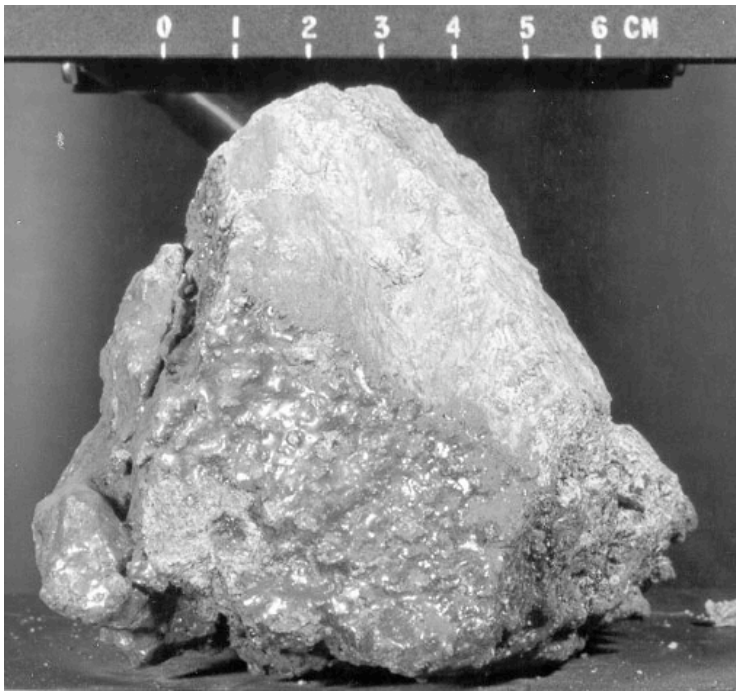
(Leonid meteoroid impact, 2006)

- **Total EVA time from the 6 Apollo missions: ~80 man-hours**

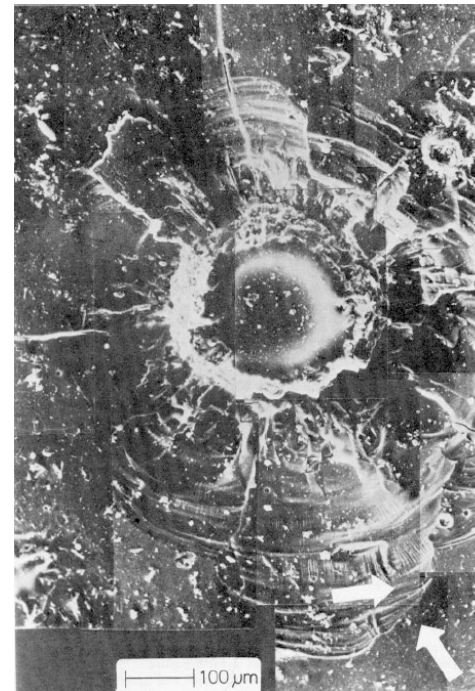


## Lunar Surface Environment (2/3)

- **Micrometeoroid (MM) flux on the surface of the Moon**
  - “Order-of-magnitude” estimates
    - The Moon:  $2 \times 10^6$  impacts/year by MM 1 cm and larger (up to  $\sim 70$  km/sec)
    - A field area the size of JSC:  $\sim 2700$  impacts/year by MM 1 mm and larger
      - @70 km/sec, a 1 mm micrometeoroid can penetrate 1.1 cm-thick Al wall



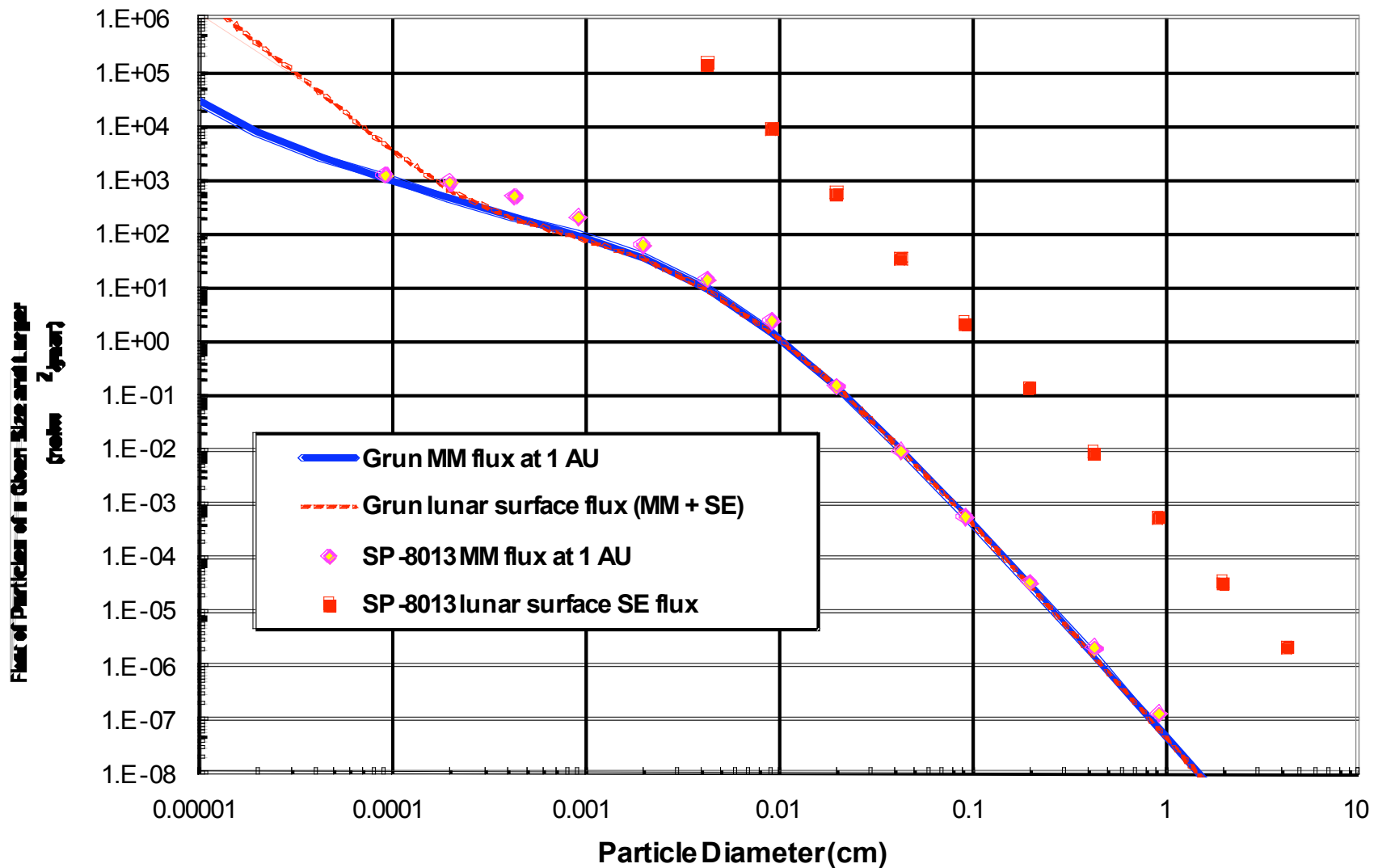
Lunar rock 12054

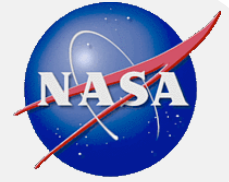




# Lunar Surface Environment (3/3)

## Micrometeoroid and Lunar Secondary Ejecta (MMSE) Models





## Benefits of the Project

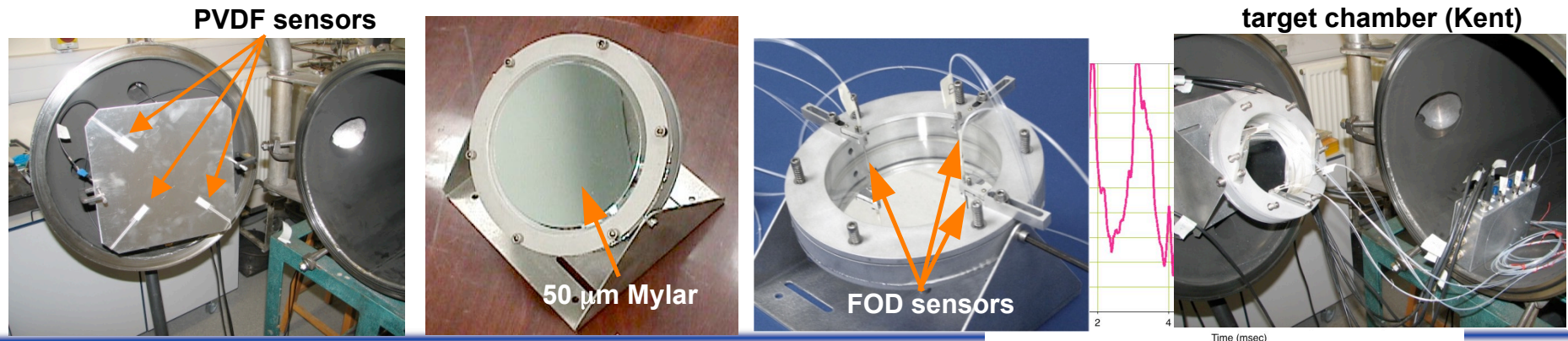
- **Fundamental Lunar Science**
  - Acquire data to improve the understanding of lunar cratering processes and the growth, mixing, and transport of the lunar regolith.
- **Lunar Exploration Applied Science**
  - Provide data for (a) conducting reliable impact risk assessments for human lunar exploration activities, (b) designing cost-effective shielding for habitats, and (c) developing mitigation measures to address dust contamination issues.
- **Planetary Science**
  - Place constraints on the collision history of asteroids, and the release of materials from comets into interplanetary space. Provide a reference to study other regolith-covered Solar System bodies from remote-sensing data.





## IMMUSE System Overview (1/2)

- **The IMMUSE system makes use of three proven particle detection technologies**
  - Impact acoustic detection
    - Using polyvinylidene fluoride (PVDF) as impact acoustic sensors
    - Development funded by the NASA PIDD Program 2003-2005
  - Fiber optic displacement (FOD) detection
    - Using FOD sensors to detect impacts on a thin film under tension
    - Funded by JSC Mission Enabling Science Program through 2012
  - Dual-layer laser curtain detection
    - Similar to the Rosetta GIADA system

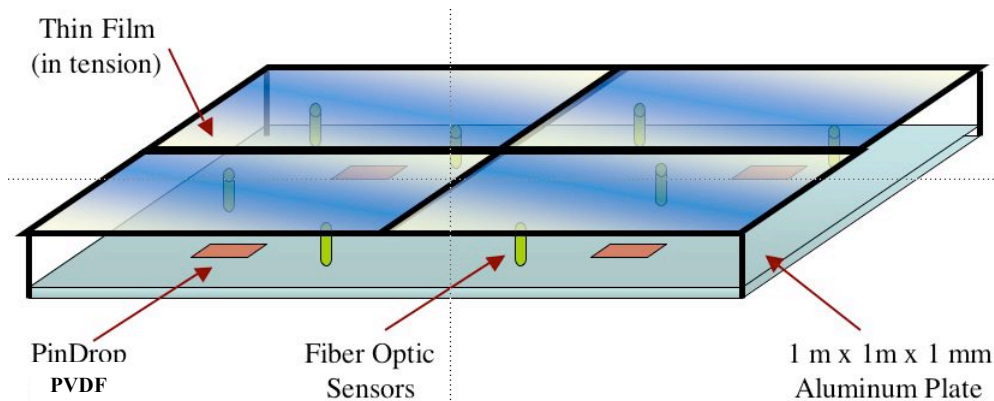






## IMMUSE System Overview (2/2)

- **IMMUSE consists of two sub-systems**
  - Micrometeoroid Impact Detection System (MIDS)
    - **FOD + PVDF sensors**
    - **The building block for a large area (>tens of m<sup>2</sup>) detector**
  - Secondary Ejecta Detection System (SEDS)
    - **Dual-layer laser curtain + PVDF sensors**
    - **10 cm × 10 cm detection area**





## Development Status

- **PVDF impact acoustic sensor**
    - Identified sensor materials for extreme temperatures and tested various backing plates
    - Conducted hypervelocity impact tests for signal characterization
  - **Fiber optic displacement sensor**
    - Designed, fabricated, and tested various configurations of the film support
    - Built prototype units for low velocity and hypervelocity impact tests
  - **Dual-layer laser curtain system**
    - Constructed a low speed particle launching device
    - Finalized the preliminary design of a single curtain system
    - Procured the parts needed for a prototype unit
- **Will continue the development of IMMUSE through 2011**



## Summary

- **Micrometeoroid and lunar secondary ejecta data are needed to better support engineering, science, and impact risk assessment applications**
- **Roadmap for the IMMUSE project**
  - Reach a Technical Readiness Level (TRL) of 3 by the end of the current LASER support in preparation for a more advanced development beyond 2011
  - Seek opportunities to collaborate with other organizations and agencies (U.S. and international) for cost-effective instrument development/advancement and potential flight opportunities to the Moon